Supplemental Type Certificate

Number: SR00763DE

This certificate issued to: Helifab, Inc.
1318 Smede Hwy
Broussard, LA 70518

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the airworthiness requirements of Part 29 of the Federal Aviation Regulations

Original Product – Type Certificate Number: Make: Sikorsky Aircraft Corporation

HINE Model: S-76A, S-76-B, S-76C, S-76D

Description of Type Design Change:


Limitations and Conditions:

1. The installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configurations will not introduce any adverse effect upon the airworthiness of the aircraft.
2. If the holder agrees to permit another person to use this certificate to alter the product, the holder must give the other person written evidence of that permission. (Limitations and Conditions continued on Page 3 of 3).

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, and revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of Application: September 27, 2012 Date Reissued: 

Date of Issuance: February 19, 2014 Date Amended: March 3, 2016

By Direction of the Administrator

Signature

John Hardie
Title Acting Manager, Fort Worth Aircraft Certification Office Southwest Region
Limitations and Conditions (Continued):

3. A copy of this certificate must be maintained as part of the permanent records for the modified rotorcraft.
4. FAA approved Rotorcraft Flight Manual Supplement, Document No. HF-S76-FMS-2800-004, Revision 1, dated November 20, 2015, or later FAA approved revision is required.
5. Instructions for Continued Airworthiness (ICA) for Auxiliary Fuel Tank Installation, Document No. HF-S76-ICA-2800-0004, Revision A, dated December 30, 2014, FAA accepted November 18, 2015, or later FAA accepted revision, must be made available to the operation at the time of installation.

Note:

It has been determined that this modification is a No Acoustical Change per 14 CFR 21.93b.

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